



# Polynomial stabilizability by optimal control.

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## **Plan**

- I Polynomial stability and stabilization**
- II Polynomial stabilization by optimal control**
- III Inverse optimal control**

This is a joint work with C. Jammazi.

## **I Polynomial stability and stabilization**

# 1 Problems in control theory

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- ▶ Observation : Reconstruct the full state of the system from partial data.



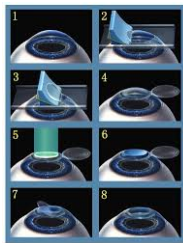
## 2 Control theory and applications



Drone



Navire



Fusée



Satellite

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$$\|x(t)\| \simeq \frac{c}{t^\alpha}, \alpha > 0.$$

- ▶ Example : the scalar system  $\dot{x} = -x^{2p+1}$ ,  $p \in \mathbb{N}^*$  is polynomially stable.  
By simple integration we have  $\|x(t)\| \leq \frac{c}{(at+b)^{\frac{1}{2p}}}$ .

## 4 Notion of polynomial stability

Let the dynamical systems

$$(1) \quad \dot{x}_1 = X_1(x_1, x_2), \dot{x}_2 = X_2(x_1, x_2),$$

$X = (X_1, X_2)$  is a continuous vector field defined on  $\mathbb{R}^p \times \mathbb{R}^{n-p}$

$x := (x_1, x_2) \in \mathbb{R}^p \times \mathbb{R}^{n-p}$  is the state.

$$(2) \quad X_1(0, x_2) = 0 \text{ and } X_2(0, x_2) = 0 \quad \forall x_2 \in \mathbb{R}^{n-p},$$

$x(0) = (x_1(0), x_2(0))$  is the initial condition.

## 5 Polynomial stability in partial sense

### Definition : Appl. Math. Comput.2013

The system (1) is said to be  $p$ -rational partially stable if the following properties are satisfied

- ▶ The origin  $(0, 0)$  of the system (1) is Lyapunov stable :

$$\forall \varepsilon > 0, \exists \eta > 0 : (|x(0)| < \eta) \Rightarrow (|x(t)| < \varepsilon \quad \forall t \geq 0).$$

- ▶ There exist positive numbers  $M, k, \eta, r$  with  $\eta \leq 1$  such that if  $|x(0)| \leq r$  then

$$\left\{ \begin{array}{l} |x_1(t)| \leq \frac{M |x(0)|^\eta}{(1 + |x(0)|^k t)^k}, \quad \forall t \geq 0, \\ \lim_{t \rightarrow +\infty} x_2(t) = a(x(0)), \end{array} \right.$$

where  $a(x(0))$  is a constant vector depending on initial conditions.

## 6 Polynomial stabilisability in partial sense

### Definition

Let be the control system

$$(3) \quad \begin{cases} \dot{x} = X(x, u) \\ X(0, 0) = 0, \end{cases}$$

where  $x \in \mathbb{R}^n$  is the state,  $u \in \mathbb{R}^m$  the control, and  $X \in C^0(\mathbb{R}^{n+m}, \mathbb{R}^n)$ . System (3) is  $p$ -partially rationally stabilizable, if there exists a continuous feedback  $x \mapsto u(x)$  such that, for every  $x_2 \in \mathbb{R}^{n-p}$ ,  $u(0, x_2) = 0$ , and  $(0, 0) \in \mathbb{R}^p \times \mathbb{R}^{n-p}$  is  $p$ -rationally stable for the closed loop system  $\dot{x} = X(x, u(x))$ .

### Remark

Obviously, the case  $n = p$  corresponds to complete rational stability of system (1).



## 7 Polynomial stabilisability in partial sense



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A. Bacciotti and L. Rosier. Liapunov Functions and Stability in Control Theory. Communications and Control Engineering, Springer-Verlag, 2005.



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## 8 Sufficient conditions for polynomial stability

### Proposition : Nonlinear Analysis-2019

Consider the dynamical system (1), we assume that there exist a locally Lipschitz function  $V : \mathbb{R}^n \rightarrow \mathbb{R}$ , some positive constants  $c_1, c_2, r_1$  and  $r_2$  such that

(a) there exists  $\varepsilon > 0$ , such that for every  $x ; |x| < \varepsilon$ ,  $V$  satisfies

$$c_1 |x|^{r_1} \leq V(x) \leq c_2 |x|^{r_2},$$

(b) there exist  $c > 0$  and  $\alpha > 0$  such that

$$(4) \quad D^+ V(x(t)) + c V^{1+\alpha}(x(t)) \leq 0.$$

Then  $0 \in \mathbb{R}^n$  is locally rationally stable. Moreover, if the first condition holds for all  $x \in \mathbb{R}^n$ , then  $0 \in \mathbb{R}^n$  is globally rationally stable.

## 9 Polynomial stability in partial sense

### Example :

$$(5) \quad \dot{x} = -x^r, \dot{y} = -yx^2,$$

where  $r \in \mathbb{Q}_{odd}^+ \cap (1, +\infty)$ .

$$\mathbb{Q}_{odd}^+ = \{r \in \mathbb{Q}_+ : r = \frac{p}{q}, \text{ where } p \text{ and } q \text{ are odd non negative integers}\}.$$

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$$V = \frac{1}{2}(x^2 + y^2), \quad \dot{V} = -x^{r+1} - x^2 y^2 \leq 0.$$

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$$W(x) = V(x, 0)$$

$\dot{W} \leq -cW^{(r+1)/2}$ , in particular

$$|x(t)| \leq \frac{k}{t^{1/(r-1)}}.$$

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If we choose,  $r < 3$  then we get  $t \mapsto x^2(t)$  is Lebesgue integrable in the neighborhood of  $+\infty$ .

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There exists  $\eta > 0$  such that  $|x(0), y(0)| < \eta$  then the integral  $\int_0^{+\infty} |y(s)x^2(s)|ds < \infty$  which implies the convergence of  $y$ .



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## II Polynomial stabilization by optimal control

- 1 Sufficient conditions for polynomial stabilization by optimal control
- 2 Application to systems with drift in Vorotnikov sense
- 3 Example

$$(6) \quad \dot{x}_1 = X_1(x, u) \dot{x}_2 = X_2(x, u)$$

$$(7) \quad X_1(0, x_2, 0) = 0 \text{ and } X_2(0, x_2, 0) = 0 \forall x_2 \in \mathbb{R}^{n-p}.$$

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$$(8) \quad J(x(0), u) = \int_0^{+\infty} L(x(t), u(t)) dt.$$

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$\Gamma(x(0)) := \{u : u \text{ is admissible, } x(t) = (x_1(t), x_2(t)) \text{ solution of (6)}$   
such that  $|x_1(t)| \leq \frac{c}{t^\alpha}$  and  $x_2(t) \rightarrow c \in \mathbb{R}^{n-p}\}.$

**Bernstein :93** Optimal feedback of nonlinear regulation problems involving a non quadratic cost functionals, the Hamilton—Jacobi—Bellman approach is used.

**Haddad et al :2014** Singular control for Linear Semistabilization

**L'Affalito et al :2015** Asymptotic stabilization (resp. finite-time stabilization) with respect to part of the system....

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### Problem

$$(\mathcal{P}) : J(x(0), u^*) = \min_{u \in \Gamma} \int_0^{+\infty} L(x(t), u(t)) dt,$$

where  $L : \mathbb{R}^n \times \mathbb{R}^m \rightarrow \mathbb{R}$  is continuous and positive function.

# 1/3 Sufficient conditions for polynomial stabilization by optimal control

## Proposition

Assume that there exist a  $\mathcal{C}^1$  function  $V : \mathcal{O} \times \mathbb{R}^{n-p} \rightarrow \mathbb{R}$ , some positive constants  $c_1, c_2, r_1, r_2, c, \alpha$ , a continuous function  $u^* : \mathcal{O} \times \mathbb{R}^{n-p} \rightarrow \mathcal{U}$  such that  $u^*(0, x_2) = 0, \forall x_2 \in \mathbb{R}^{n-p}$  and

(a)

$$(9) \quad c_1 |x_1|^{r_1} \leq V(x) \leq c_2 |x_1|^{r_2}, \forall x_1 \in \mathcal{O},$$

$$(10) \quad \dot{V}(x) \leq -c V^{\alpha+1},$$

$$(11) \quad (\exists \eta > 0 : |x(0)| < \eta) \Rightarrow \int_0^{+\infty} |X_2(x(s), u^*(x(s)))| ds < \infty,$$



## 1/4 Sufficient conditions for polynomial stabilization by optimal control

(b) for every  $x = (x_1, x_2) \in \mathcal{O} \times \mathbb{R}^{n-p} : V$  and  $L$  satisfy

$$(12) \quad L(x, u^*) + \nabla V(x) \cdot X(x, u^*) = 0$$

$$(13) \quad L(x, u) + \nabla V(x) \cdot X(x, u) \geq 0.$$

Then, the system (6) is  $p$ -partially locally rationally stabilizable, and

$$(14) \quad J(x(0), u^*) = V(x(0)).$$

Furthermore the feedback control  $u^*$  minimizes  $J$  in the sense that

$$(15) \quad J(x(0), u^*) = \min_{u \in \Gamma(x(0))} J(x(0), u).$$

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$$u^* := \operatorname{argmin}_{u \in \Gamma(x(0))} [L(x, u) + \nabla V(x) \cdot X(x, u)].$$

## II Polynomial stabilization by optimal control

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$$(16) \quad \begin{cases} \dot{x}_1 &= f_1(x) + G_1(x) u \\ \dot{x}_2 &= f_2(x) + G_2(x) u, \end{cases}$$

where  $x = (x_1, x_2) \in \mathbb{R}^p \times \mathbb{R}^{n-p}$ ,  $u \in \mathbb{R}^m$ ,  $f = (f_1, f_2)$  defined on  $\mathbb{R}^p \times \mathbb{R}^{n-p}$  and  $G = (G_1, G_2) : \mathbb{R}^p \times \mathbb{R}^{n-p} \rightarrow \mathbb{R}^{p \times (n-p) \times m}$ .

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$$(17) \quad J(x(0), u) = \int_0^{+\infty} L(x, u(t)) dt.$$

$$(18) \quad L(x, u) = L_1(x) + L_2(x) u + u^T R u,$$

where  $R : \mathbb{R}^m \rightarrow \mathbb{R}^{m \times m}$  is positive definite matrix-valued function.

$\Lambda(x(0)) := \{u : u \text{ is admissible, } x(t) = (x_1(t), x_2(t))$   
 solution of (16) and  $|x_1(t)| \leq \frac{c}{t^\alpha}\}$ .

### Proposition

Consider the system with drift (16) with cost functional (17). We assume that there exist a  $\mathcal{C}^1$ -function  $V : \mathbb{R}^p \times \mathbb{R}^{n-p} \rightarrow \mathbb{R}$  and some positive constants  $c_1, c_2, r_1, r_2, r_3, c, \alpha$  such that

(a) for every  $x \in \mathbb{R}^n$ ,

$$(19) \quad c_1 |x_1|^{r_1} \leq V(x) \leq c_2 |x_1|^{r_2},$$

$$(20) \quad L_f V(x) - \frac{1}{2} L_G V(x) R^{-1} L_G^T(x) - \frac{1}{2} L_G V(x) R^{-1} (L_G V(x))^T \leq -c V^{\alpha+1}$$

(b)

$$(21) \quad L_2(0, x_2) = 0, \quad \forall x_2 \in \mathbb{R}^{n-p}$$

## 2/3 Application to systems with drift in Vorotnikov sense

$$(22) \quad L_1(x) + L_f V(x) - \frac{1}{4} [L_G V(x) + L_2(x)] \cdot R^{-1} [L_G V(x) + L_2(x)]^T = 0.$$

Then the system (16) is  $p$ -partially rationally stable with Vorotnikov sense under the optimal feedback

$$(23) \quad u^* := -\frac{1}{2} R^{-1} [L_2(x) + L_G V(x)]^T.$$

In addition,  $u^*$  minimizes the cost functional  $J$  in the sense that

$$(24) \quad J(x(0), u^*) = \min_{u \in \Lambda(x(0))} J(x(0), u),$$

and

$$(25) \quad J(x(0), u^*) = V(x(0)).$$

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The model of spacecraft with two axis :

$$(26) \quad \begin{cases} \dot{\omega}_1 &= \alpha_1 u_1 \\ \dot{\omega}_2 &= \alpha_1 u_2 \\ \dot{\omega}_3 &= \alpha_3 u_1 + \alpha_4 u_2, \end{cases}$$

where

- ▶  $x = (\omega_i)_{1 \leq i \leq 3}^T \in \mathbb{R}^3$  is the state and denote the components of the angular velocity vector with respect to a given inertial reference frame expressed in a central body reference frame,
- ▶  $u_1, u_2$  are the spacecraft control moments,

### 3/2 Example

$$(27) \quad J(x(0), u) = \int_0^{+\infty} \frac{1}{16} \alpha_1^2 |x_1|^{\frac{4k}{p}+2}, \quad p, k \in \mathbb{R}^*,$$

### 3/2 Example

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$$V(x) = \frac{p}{4(k+p)} |x_1|^{\frac{2k}{p}+2}, \quad p, k \in \mathbb{R}_+^*.$$

$$(28) \quad \begin{cases} u_1^*(x) &= -\frac{1}{4} \alpha_1 |x_1|^{\frac{2k}{p}} \omega_1 \\ u_2^*(x) &= -\frac{1}{4} \alpha_1 |x_1|^{\frac{2k}{p}} \omega_2, \end{cases}$$

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$$(29) \quad |\omega_i(t)| \leq \frac{\sqrt{c_1}}{t^{\frac{p}{2k}}}, \quad i = 1, 2,$$

if  $p > 2k$ , then the solutions  $\omega_i(t)$ ,  $i = 1, 2$  are Lebesgue-integrable, then the state  $\omega_3(t)$  converges.

## Inverse optimal control

# 1 Inverse optimal control

- Problem :

Given a control system,

$$\dot{x} = f(x, u),$$

a specific feedback law control  $u^*$  stabilizes this system, with respect to a positive definite radially unbounded Lyapunov function  $V$ .

# 1 Inverse optimal control

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Given a control system,

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a specific feedback law control  $u^*$  stabilizes this system, with respect to a positive definite radially unbounded Lyapunov function  $V$ .

- Goal :

to find a Lagrangian function  $L$  for which this control  $u^*$  is optimal in integral cost sense.

## 2 Inverse optimal control

1. **Freeman et Kokotovic :96** Optimalité inverse pour la stabilisation robuste.
2. **Tsiotras :99** Optimalité inverse pour la stabilisation d'un satellite.
3. **Edouard et al :2014** Problème de contrôle optimal inverse avec une optimisation polynomiale.
4. **Haddad :2014, L'Afflitto al :2015, 2016** Stabilisation asymptotique (respectivement en temps fini) par rapport à une partie du système par un contrôle optimal inverse...



### 3 Inverse optimal control

#### Proposition

Consider the system with drift (16) subject to cost functional (17). We assume that there exist a  $\mathcal{C}^1$ -function  $V$  and some positive constants  $c_1, c_2, r_1, r_2, r_3, c, \alpha$  such that the set of conditions hold

(a)

$$c_1 |x|^{r_1} \leq V(x) \leq c_2 |x|^{r_2},$$

for every  $x \in \mathbb{R}^n$

$$L_f V(x) - \frac{1}{2} L_G V(x) R^{-1} L_G^T V(x) - \frac{1}{2} L_G V(x) R^{-1} (L_G V(x))^T \leq -c V^{\alpha+1}(x)$$

(b)

$$L_2(0, x_2) = 0, \forall x_2 \in \mathbb{R}^{n-p}$$

then under the feedback

## 4 Inverse optimal control

$$u^* := -\frac{1}{2} R^{-1} \cdot [L_2(x) + L_G V(x)]^T,$$

the closed loop system (16) is partially rationally stable in Vorotnikov sense. Furthermore  $u^*$  is optimal with respect the cost functional (17), where

$$L_1(x) = u^{*T}(x) R \cdot u^*(x) - L_f V(x)$$

i.e.  $J$  is minimized in the sense (24) and (25).

## 5 Example

The model of spacecraft with one axe of symmetry :

$$(30) \quad \begin{cases} \dot{\omega}_1 &= I_{23} \omega_2 \omega_3 + u_1 \\ \dot{\omega}_2 &= -I_{23} \omega_3 \omega_1 + u_2 \\ \dot{\omega}_3 &= \alpha_3 u_1 + \alpha_4 u_2, \end{cases}$$

- ▶  $x := (\omega_i)_{1 \leq i \leq 3}^T \in \mathbb{R}^3$  is the state and denote the components of the angular velocity vector with respect to a given inertial reference.
- ▶  $u_1, u_2$  are the spacecraft control moments.
- ▶  $\alpha_3, \alpha_4 \in \mathbb{R}$ .
- ▶  $I_{23} = \frac{I_2 - I_3}{I_1}$ ,  $I_1, I_2$  and  $I_3$  are the principal moments of inertia of the spacecraft such that  $0 < I_1 = I_2 < I_3$ .

## 6 Example

$$(31) \quad V(x) = \frac{p}{2(p+k)} |x_1|^{\frac{2k}{p}+2}, \quad p, k \in \mathbb{R}_+^*.$$

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## 6 Example

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$$L_1(x) = u^{*T}(x) R(x) u^*(x) - V'(x) f(x) = (I_{23} \omega_2 \omega_3)^2 + (I_{23} \omega_3 \omega_1)^2 + \frac{1}{2} |x_1|^{4\frac{k}{p}+2},$$

$$J(x(0), u) = \int_0^{+\infty} L(x, u(t)) ds.$$

Thank you for your attention

